

## SUPPLEMENTAL TYPE CERTIFICATE

**10062665**

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to:

### AIRBUS HELICOPTERS CANADA LIMITED

1100 GILMORE ROAD  
FORT ERIE ON L2A 5M9  
CANADA

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

**Original Type Certificate Number:** EASA.R.009

**Type Certificate Holder:** AIRBUS HELICOPTERS DEUTSCHLAND

**Type:** EC135

**Model:** EC135 P1, EC135 P2, EC135 P3

EC135 P2+, EC135 T2+

EC135 T1, EC135 T2, EC135 T3

**Original STC Number:** TCCA SH16-14 ISSUE 1

**Description of Design Change:**

Installation of Removable Pins for Hinged Door in accordance with Master Drawing List MDL-14-005.

See Continuation Sheet(s)

For the European Aviation Safety Agency

Date of Issue: 20 July 2017



**Pier Giorgio COLOMBO**  
Medium Rotorcraft Section  
Manager

10043485

SUPPLEMENTAL TYPE CERTIFICATE - 10062665 - AIRBUS HELICOPTERS CANADA LIMITED - 301635



**EASA Certification Basis:**

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.  
The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

**Associated Technical Documentation:**

- Master Drawing List MDL-14-005, Issue C.
- Instructions for Continued Airworthiness ICA-AHCA-295, Revision 0.

**Limitations/Conditions:**

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -