1. **AFFECTED AIRCRAFT:**
   SIAT 223, (EASA.A.554) all Serial Numbers

2. **TIME OF COMPLIANCE:**
   Before next flight, thereafter every 12 months

3. **SUBJECT:**
   Inspection of bearing bolt on R/H and L/H flap

4. **REASON:**
   Crack found on weld (bearing bolt / fitting)

5. **WEIGHT AND BALANCE:**
   Not affected

6. **REQUIRED MATERIAL:**
   None

7. **REMARKS:**
   A cracked bearing bolt weld was found on a flap during an inspection (see sketch / photo).

   Several flaps were additionally inspected. Several repaired welds were found.
8. **IMPLEMENTING MEASURES:**

The welded areas of both bearing bolts on the L/H and R/H flap have to be cleaned and visually inspected.

If cracks are found in the welding area, these have to be repaired prior to the next flight. If the inner flap fitting cannot be repaired, the entire inner fitting has to be replaced.

The technical content of this document is approved under the authority of EASA Design Organization Approval No. EASA.21J.026.

**Genehmigt / Approved:**

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